

## 1

### Stress 3



- Chapter Objectives 3
- 1.1 Introduction 3
- 1.2 Equilibrium of a Deformable Body 4
- 1.3 Stress 22
- 1.4 Average Normal Stress in an Axially Loaded Bar 24
- 1.5 Average Shear Stress 32
- 1.6 Allowable Stress Design 46
- 1.7 Limit State Design 48



✓ **External Loads.** A body is subjected to only two types of external loads; namely, surface forces and body forces, Fig. 1–1 .

❖ **Surface forces** :- are caused by the direct contact of one body with the surface of another.

- *concentrated force*
- *distributed load* ( measured by their intensity).
  - Uniform load ( Rectangular load, triangular load)
  - Linear distributed load.

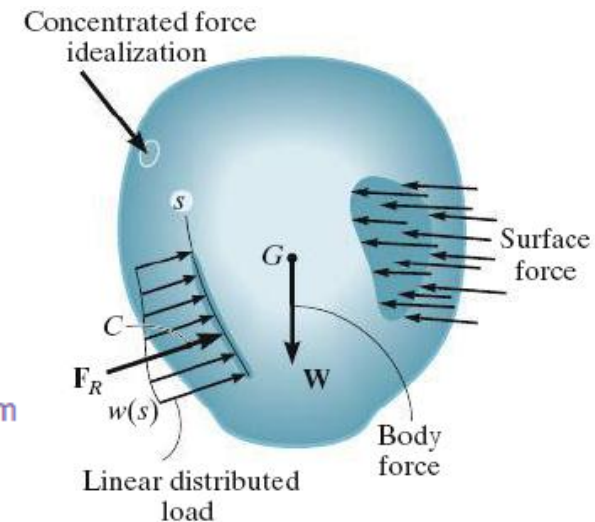
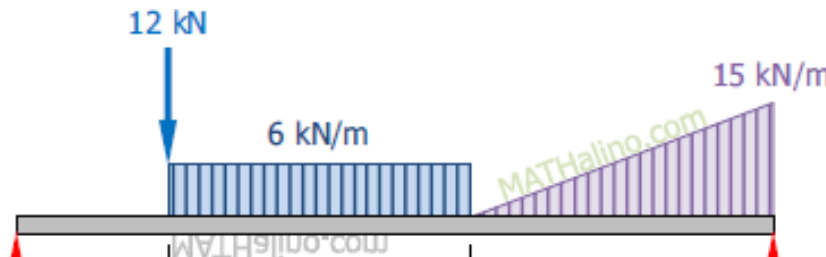
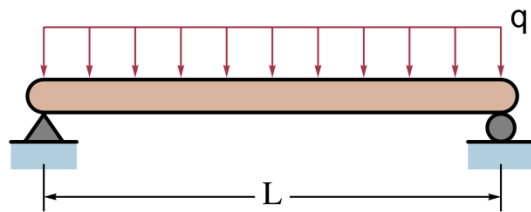


Fig. 1-1

## Equations of Equilibrium.

Equilibrium of a body requires both a ***balance of forces***, to prevent the body from translating or having accelerated motion along a straight or curved path, and a ***balance of moments***, to prevent the body from rotating. These conditions can be expressed mathematically by two vector equations.

### Equilibrium of a body:

$$\begin{array}{lll} \Sigma F_x = 0 & \Sigma F_y = 0 & \Sigma F_z = 0 \\ \Sigma M_x = 0 & \Sigma M_y = 0 & \Sigma M_z = 0 \end{array} \quad (1-2)$$

**For coplanar forces :** There are three equations of equilibrium

$$\begin{array}{l} \Sigma F_x = 0 \\ \Sigma F_y = 0 \\ \Sigma M_o = 0 \end{array} \quad (1-3)$$

## ✓ Internal loadings:

1. These internal loading acting on a specific region within the body can be attained by the *Method of Section*.
2. *Method of Section*: Imaginary cut is made through the body in the region where the internal loading is to be determined.
3. The two parts are separated and a free body diagram of one of the parts is drawn. Point O is often chosen as the centroid of the sectioned area

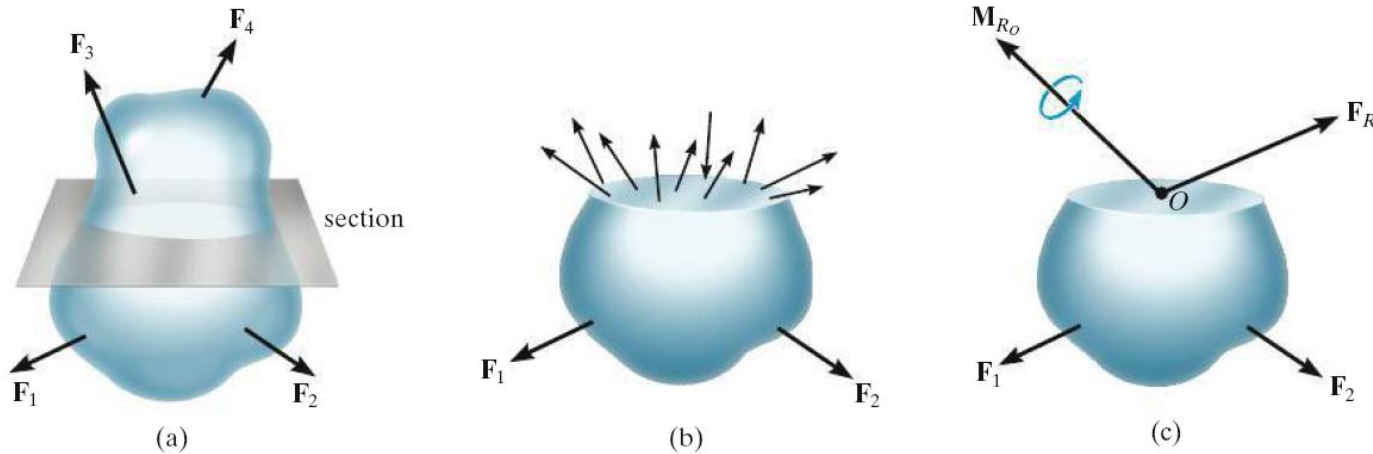
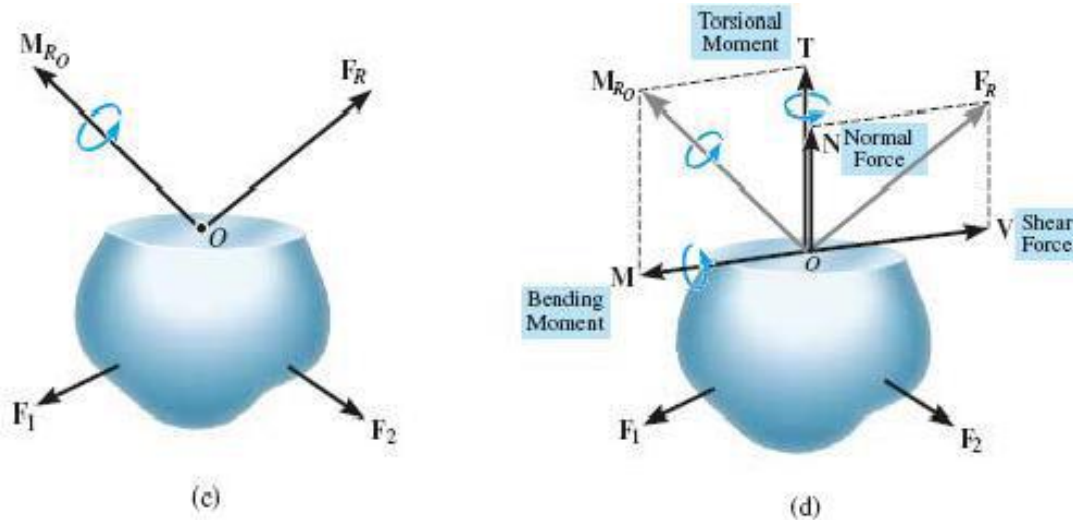


Fig. 1-2

Apply Equilibrium at this stage

Four types of internal loadings can be defined:



**Normal force (N).** This force act perpendicular to the area.

**Shear Force (V).** This force lies in the plane of the area (parallel) **Torsional Moment (T).** This torque is developed when the external loads tend to twist one segment of the body with respect to the other

**Bending Moment (M).** This moment is developed when the external loads tend to bend the body.

# STRESS

6

**Ex1:-** Determine the resultant internal loadings acting on the cross section at *C* of the cantilevered beam shown in Fig. 1–4 *a*.

Determine the internal loading at C A cut will be made through C and the right part will be studied.

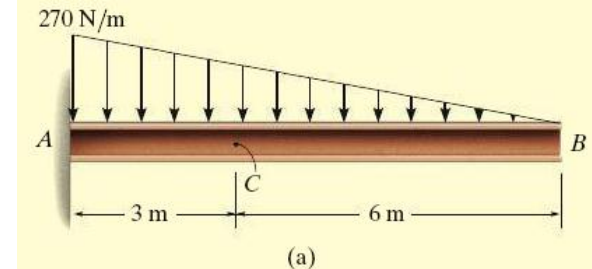


Fig. 1–4

**Equations of Equilibrium.** Applying the equations of equilibrium we have

$$\pm \rightarrow \Sigma F_x = 0;$$

$$-N_C = 0$$

$$N_C = 0$$

*Ans.*

$$+ \uparrow \Sigma F_y = 0;$$

$$V_C - 540 \text{ N} = 0$$

$$V_C = 540 \text{ N}$$

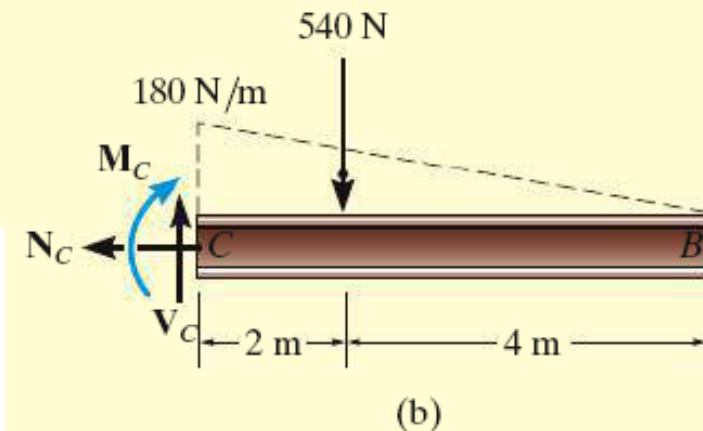
*Ans.*

$$\zeta + \Sigma M_C = 0;$$

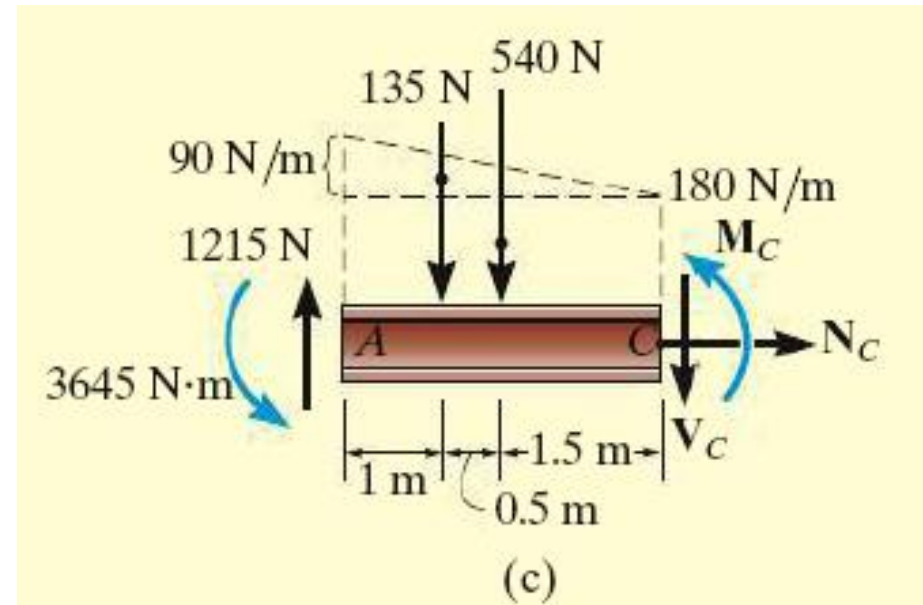
$$-M_C - 540 \text{ N}(2 \text{ m}) = 0$$

$$M_C = -1080 \text{ N} \cdot \text{m}$$

*Ans.*



If the cut was made at C and the left part was taken First a free body diagram for the entire body is made and equilibrium is applied to get the support reactions.



**Ex2:-** The **500-kg** engine is suspended from the crane boom in Fig. 1–5 *a*. Determine the resultant internal loadings acting on the cross section of the boom at point *E*.

$$\zeta + \Sigma M_A = 0; \quad F_{CD} \left( \frac{3}{5} \right) (2 \text{ m}) - [500(9.81) \text{ N}](3 \text{ m}) = 0$$

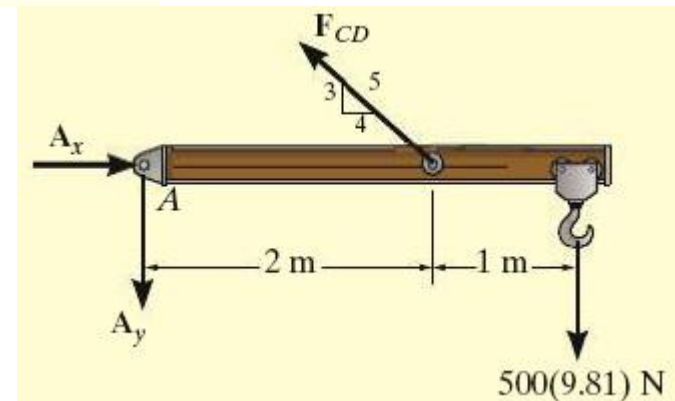
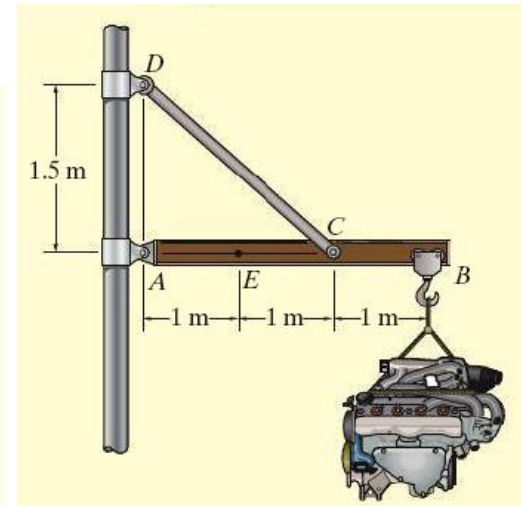
$$F_{CD} = 12\,262.5 \text{ N}$$

$$\pm \Sigma F_x = 0; \quad A_x - (12\,262.5 \text{ N}) \left( \frac{4}{5} \right) = 0$$

$$A_x = 9810 \text{ N}$$

$$+ \uparrow \Sigma F_y = 0; \quad -A_y + (12\,262.5 \text{ N}) \left( \frac{3}{5} \right) - 500(9.81) \text{ N} = 0$$

$$A_y = 2452.5 \text{ N}$$





**Free-Body Diagram.** The free-body diagram of segment  $AE$  is shown in Fig. 1–5  $c$ .

**Equations of Equilibrium.**

$$\pm \rightarrow \Sigma F_x = 0; \quad N_E + 9810 \text{ N} = 0$$

$$N_E = -9810 \text{ N} = -9.81 \text{ kN}$$

$$+\uparrow \Sigma F_y = 0; \quad -V_E - 2452.5 \text{ N} = 0$$

$$V_E = -2452.5 \text{ N} = -2.45 \text{ kN}$$

$$\zeta + \Sigma M_E = 0; \quad M_E + (2452.5 \text{ N})(1 \text{ m}) = 0$$

$$M_E = -2452.5 \text{ N} \cdot \text{m} = -2.45 \text{ kN} \cdot \text{m}$$

