

# State-space methods for control system design

## 8.1 The state-space-approach

The classical control system design techniques discussed in Chapters 5–7 are generally only applicable to

- (a) Single Input, Single Output (SISO) systems
- (b) Systems that are linear (or can be linearized) and are time invariant (have parameters that do not vary with time).

The state-space approach is a generalized time-domain method for modelling, analysing and designing a wide range of control systems and is particularly well suited to digital computational techniques. The approach can deal with

- (a) Multiple Input, Multiple Output (MIMO) systems, or multivariable systems
- (b) Non-linear and time-variant systems
- (c) Alternative controller design approaches.

### 8.1.1 The concept of state

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The state of a system may be defined as: ‘The set of variables (called the state variables) which at some initial time  $t_0$ , together with the input variables completely determine the behaviour of the system for time  $t \geq t_0$ ’.

The state variables are the smallest number of states that are required to describe the dynamic nature of the system, and it is not a necessary constraint that they are measurable. The manner in which the state variables change as a function of time may be thought of as a trajectory in  $n$  dimensional space, called the *state-space*. Two-dimensional state-space is sometimes referred to as the *phase-plane* when one state is the derivative of the other.

## 8.1.2 The state vector differential equation

The state of a system is described by a set of first-order differential equations in terms of the state variables  $(x_1, x_2, \dots, x_n)$  and input variables  $(u_1, u_2, \dots, u_m)$  in the general form

$$\begin{aligned}\frac{dx_1}{dt} &= a_{11}x_1 + a_{12}x_2 + \dots + a_{1n}x_n + b_{11}u_1 + \dots + b_{1m}u_m \\ \frac{dx_2}{dt} &= a_{21}x_1 + a_{22}x_2 + \dots + a_{2n}x_n + b_{21}u_1 + \dots + b_{2m}u_m \\ \frac{dx_n}{dt} &= a_{n1}x_1 + a_{n2}x_2 + \dots + a_{nn}x_n + b_{n1}u_1 + \dots + b_{nm}u_m\end{aligned}\quad (8.1)$$

The equations set (8.1) may be combined in matrix format. This results in the state vector differential equation

$$\dot{\mathbf{x}} = \mathbf{Ax} + \mathbf{Bu} \quad (8.2)$$

Equation (8.2) is generally called the state equation(s), where lower-case boldface represents vectors and upper-case boldface represents matrices. Thus

$\mathbf{x}$  is the  $n$  dimensional state vector

$$\begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix} \quad (8.3)$$

$\mathbf{u}$  is the  $m$  dimensional input vector

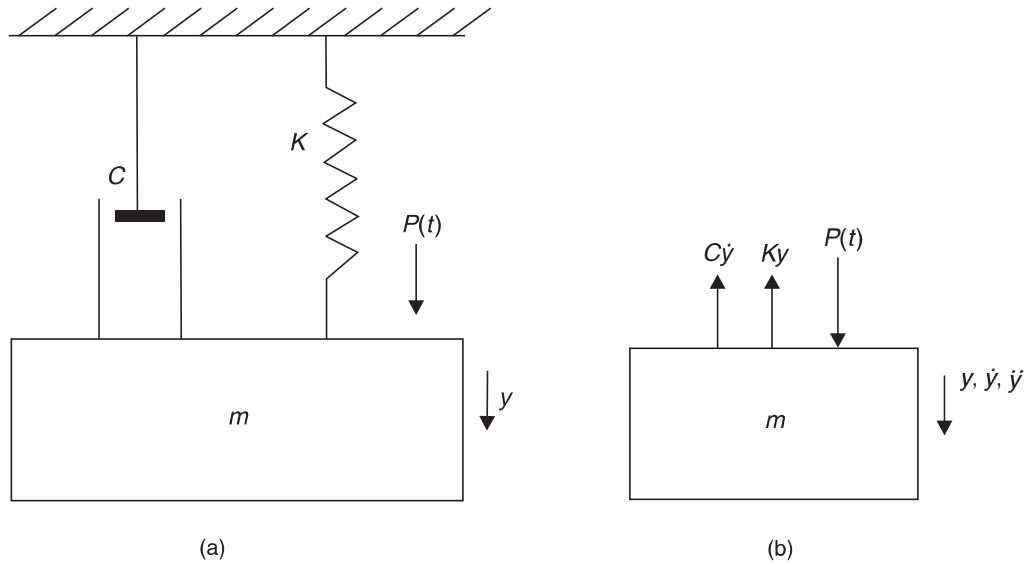
$$\begin{bmatrix} u_1 \\ u_2 \\ \vdots \\ u_m \end{bmatrix} \quad (8.4)$$

$\mathbf{A}$  is the  $n \times n$  system matrix

$$\begin{bmatrix} a_{11} & a_{12} & \dots & a_{1n} \\ a_{21} & a_{22} & \dots & a_{2n} \\ \vdots & & & \\ a_{n1} & a_{n2} & \dots & a_{nn} \end{bmatrix} \quad (8.5)$$

$\mathbf{B}$  is the  $n \times m$  control matrix

$$\begin{bmatrix} b_{11} & \dots & b_{1m} \\ b_{21} & \dots & b_{2m} \\ \vdots & & \\ b_{n1} & \dots & b_{nm} \end{bmatrix} \quad (8.6)$$



**Fig. 8.1** Spring–mass–damper system and free-body diagram.

In general, the outputs ( $y_1, y_2, \dots, y_n$ ) of a linear system can be related to the state variables and the input variables

$$\mathbf{y} = \mathbf{C}\mathbf{x} + \mathbf{D}\mathbf{u} \quad (8.7)$$

Equation (8.7) is called the output equation(s).

*Example 8.1*

Write down the state equation and output equation for the spring–mass–damper system shown in Figure 8.1(a).

*Solution*

State variables

$$x_1 = y \quad (8.8)$$

$$x_2 = \frac{dy}{dt} = \dot{x}_1 \quad (8.9)$$

Input variable

$$u = P(t) \quad (8.10)$$

Now

$$\sum F_y = m\ddot{y}$$

From Figure 8.1(b)

$$P(t) - Ky - C\dot{y} = m\ddot{y}$$

or

$$\frac{d^2y}{dt^2} = -\frac{K}{m}y - \frac{C}{m}\dot{y} + \frac{1}{m}P(t) \quad (8.11)$$

From equations (8.9), (8.10) and (8.11) the set of first-order differential equations are

$$\begin{aligned} \dot{x}_1 &= x_2 \\ \dot{x}_2 &= -\frac{K}{m}x_1 - \frac{C}{m}x_2 + \frac{1}{m}u \end{aligned} \quad (8.12)$$

and the state equations become

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -\frac{K}{m} & -\frac{C}{m} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{m} \end{bmatrix} u \quad (8.13)$$

From equation (8.8) the output equation is

$$y = [1 \quad 0] \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} \quad (8.14)$$

State variables are not unique, and may be selected to suit the problem being studied.

### Example 8.2

For the *RCL* network shown in Figure 8.2, write down the state equations when

- (a) the state variables are  $v_2(t)$  and  $\dot{v}_2$
- (b) the state variables are  $v_2(t)$  and  $i(t)$ .

### Solution

(a)

$$\begin{aligned} x_1 &= v_2(t) \\ x_2 &= \dot{v}_2 = \dot{x}_1 \end{aligned} \quad (8.15)$$

From equation (2.37)

$$LC \frac{d^2 v_2}{dt^2} + RC \frac{dv_2}{dt} + v_2 = v_1(t) \quad (8.16)$$

From equations (8.15) and (8.16) the set of first-order differential equations are

$$\begin{aligned} \dot{x}_1 &= x_2 \\ \dot{x}_2 &= -\frac{1}{LC}x_1 - \frac{RC}{LC}x_2 + \frac{1}{LC}u \end{aligned} \quad (8.17)$$

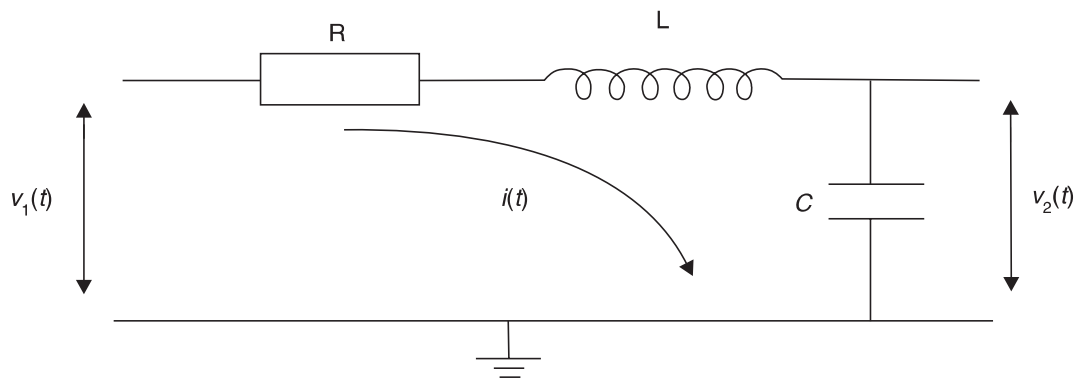


Fig. 8.2 *RCL* network.

and the state equations are

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -\frac{1}{LC} & -\frac{R}{L} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{LC} \end{bmatrix} u \quad (8.18)$$

$$(b) \quad \begin{aligned} x_1 &= v_2(t) \\ x_2 &= i(t) \end{aligned} \quad (8.19)$$

From equations (2.34) and (2.35)

$$L \frac{di}{dt} = -v_2(t) - Ri(t) + v_1(t) \quad (8.20)$$

$$C \frac{dv_2}{dt} = i(t) \quad (8.21)$$

Equations (8.20) and (8.21) are both first-order differential equations, and can be written in the form

$$\begin{aligned} \dot{x}_1 &= \frac{1}{C} x_2 \\ \dot{x}_2 &= -\frac{1}{L} x_1 - \frac{R}{L} x_2 + \frac{1}{L} u \end{aligned} \quad (8.22)$$

giving the state equations

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -\frac{1}{L} & -\frac{R}{L} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{L} \end{bmatrix} u \quad (8.23)$$

### *Example 8.3*

For the 2 mass system shown in Figure 8.3, find the state and output equation when the state variables are the position and velocity of each mass.

#### *Solution*

State variables

$$\begin{aligned} x_1 &= y_1 & x_2 &= \dot{y}_1 \\ x_3 &= y_2 & x_4 &= \dot{y}_2 \end{aligned}$$

System outputs

$$y_1, y_2$$

System inputs

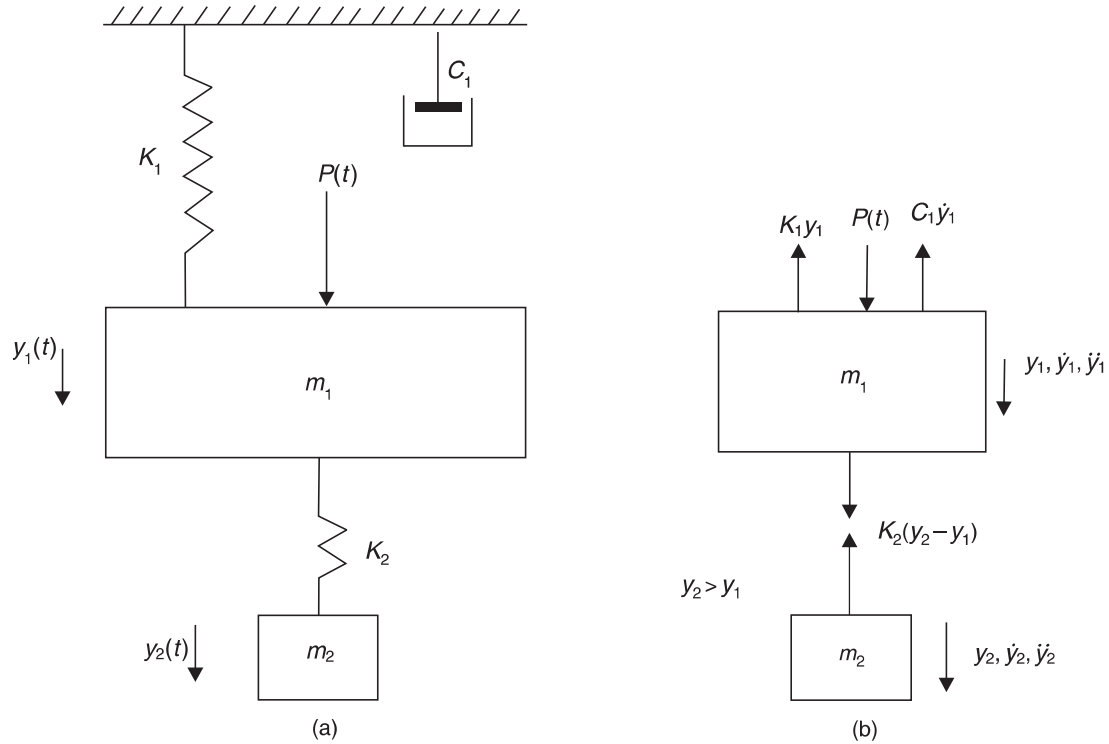
$$u = P(t) \quad (8.24)$$

For mass  $m_1$

$$\begin{aligned} \sum F_y &= m_1 \ddot{y}_1 \\ K_2(y_2 - y_1) - K_1 y_1 + P(t) - C_1 \dot{y}_1 &= m_1 \ddot{y}_1 \end{aligned} \quad (8.25)$$

For mass  $m_2$

$$\begin{aligned} \sum F_y &= m_2 \ddot{y}_2 \\ -K_2(y_2 - y_1) &= m_2 \ddot{y}_2 \end{aligned} \quad (8.26)$$



**Fig. 8.3** Two-mass system and free-body diagrams.

From (8.24), (8.25) and (8.26), the four first-order differential equations are

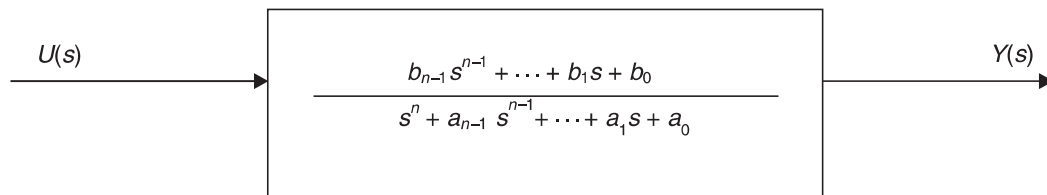
$$\begin{aligned}
 \dot{x}_1 &= x_2 \\
 \dot{x}_2 &= \left(-\frac{K_1}{m_1} - \frac{K_2}{m_1}\right)x_1 - \frac{C_1}{m_1}x_2 + \frac{K_2}{m_1}x_3 + \frac{1}{m_1}u \\
 \dot{x}_3 &= x_4 \\
 \dot{x}_4 &= \frac{K_2}{m_2}x_1 - \frac{K_2}{m_2}x_3
 \end{aligned} \tag{8.27}$$

Hence the state equations are

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \dot{x}_4 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ -\left(\frac{K_1 + K_2}{m_1}\right) & -\frac{C_1}{m_1} & \frac{K_2}{m_1} & 0 \\ 0 & 0 & 0 & 1 \\ \frac{K_2}{m_2} & 0 & -\frac{K_2}{m_2} & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{m_1} \\ 0 \\ 0 \end{bmatrix} u \tag{8.28}$$

and the output equations are

$$\begin{bmatrix} y_1 \\ y_2 \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} \tag{8.29}$$

**Fig. 8.4** Generalized transfer function.

### 8.1.3 State equations from transfer functions

Consider the general differential equation

$$\frac{d^n y}{dt^n} + a_{n-1} \frac{d^{n-1} y}{dt^{n-1}} + \dots + a_1 \frac{dy}{dt} + a_0 y = b_{n-1} \frac{d^{n-1} u}{dt^{n-1}} + \dots + b_1 \frac{du}{dt} + b_0 u \quad (8.30)$$

Equation (8.30) can be represented by the transfer function shown in Figure 8.4.

Define a set of state variables such that

$$\begin{aligned} \dot{x}_1 &= x_2 \\ \dot{x}_2 &= x_3 \\ &\vdots \\ \dot{x}_n &= -a_0 x_1 - a_1 x_2 - \dots - a_{n-1} x_n + u \end{aligned} \quad (8.31)$$

and an output equation

$$y = b_0 x_1 + b_1 x_2 + \dots + b_{n-1} x_n \quad (8.32)$$

Then the state equation is

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \vdots \\ \dot{x}_{n-1} \\ \dot{x}_n \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & \dots & 0 \\ 0 & 0 & 1 & \dots & 0 \\ \vdots & & & & \\ 0 & 0 & 0 & \dots & 1 \\ -a_0 & -a_1 & -a_2 & \dots & -a_{n-1} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_{n-1} \\ x_n \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ \vdots \\ 0 \\ 1 \end{bmatrix} u \quad (8.33)$$

The state-space representation in equation (8.33) is called the controllable canonical form and the output equation is

$$y = [b_0 \quad b_1 \quad b_2 \quad \dots \quad b_{n-1}] \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ \vdots \\ x_n \end{bmatrix} \quad (8.34)$$

*Example 8.4* (See also Appendix 1, *examp84.m*)

Find the state and output equations for

$$\frac{Y}{U}(s) = \frac{4}{s^3 + 3s^2 + 6s + 2}$$

*Solution*

State equation

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -2 & -6 & -3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} u \quad (8.35)$$

Output equation

$$y = [4 \quad 0 \quad 0] \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} \quad (8.36)$$

*Example 8.5*

Find the state and output equations for

$$\frac{Y}{U}(s) = \frac{5s^2 + 7s + 4}{s^3 + 3s^2 + 6s + 2}$$

*Solution*

The state equation is the same as (8.35). The output equation is

$$y = [4 \quad 7 \quad 5] \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} \quad (8.37)$$

## 8.2 Solution of the state vector differential equation

Consider the first-order differential equation

$$\frac{dx}{dt} = ax(t) + bu(t) \quad (8.38)$$

where  $x(t)$  and  $u(t)$  are scalar functions of time. Take Laplace transforms

$$sX(s) - x(0) = aX(s) + bU(s) \quad (8.39)$$

where  $x(0)$  is the initial condition. From equation (8.39)

$$X(s) = \frac{x(0)}{(s-a)} + \frac{b}{(s-a)} U(s) \quad (8.40)$$

Inverse transform

$$x(t) = e^{at}x(0) + \int_0^t e^{a(t-\tau)}bu(\tau)d\tau \quad (8.41)$$

where the integral term in equation (8.41) is the convolution integral and  $\tau$  is a dummy time variable. Note that

$$e^{at} = 1 + at + \frac{a^2t^2}{2!} + \cdots + \frac{a^k t^k}{k!} \quad (8.42)$$



Consider now the state vector differential equation

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u} \quad (8.43)$$

Taking Laplace transforms

$$s\mathbf{X}(s) - \mathbf{x}(0) = \mathbf{A}\mathbf{X}(s) + \mathbf{B}\mathbf{U}(s) \quad (8.44)$$

$$(s\mathbf{I} - \mathbf{A})\mathbf{X}(s) = \mathbf{x}(0) + \mathbf{B}\mathbf{U}(s)$$

Pre-multiplying by  $(s\mathbf{I} - \mathbf{A})^{-1}$

$$\mathbf{X}(s) = (s\mathbf{I} - \mathbf{A})^{-1}\mathbf{x}(0) + (s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B}\mathbf{U}(s) \quad (8.45)$$

Inverse transform

$$\mathbf{x}(t) = e^{\mathbf{A}t}\mathbf{x}(0) + \int_0^t e^{\mathbf{A}(t-\tau)}\mathbf{B}\mathbf{u}(\tau)d\tau \quad (8.46)$$

if the initial time is  $t_0$ , then

$$\mathbf{x}(t) = e^{\mathbf{A}(t-t_0)}\mathbf{x}(0) + \int_{t_0}^t e^{\mathbf{A}(t-\tau)}\mathbf{B}\mathbf{u}(\tau)d\tau \quad (8.47)$$

The exponential matrix  $e^{\mathbf{A}t}$  in equation (8.46) is called the state-transition matrix  $\Phi(t)$  and represents the natural response of the system. Hence

$$\Phi(s) = (s\mathbf{I} - \mathbf{A})^{-1} \quad (8.48)$$

$$\Phi(t) = \mathcal{L}^{-1}(s\mathbf{I} - \mathbf{A})^{-1} = e^{\mathbf{A}t} \quad (8.49)$$

Alternatively

$$\Phi(t) = \mathbf{I} + \mathbf{A}t + \frac{\mathbf{A}^2 t^2}{2!} + \cdots + \frac{\mathbf{A}^k t^k}{k!} \quad (8.50)$$

Hence equation (8.46) can be written

$$\mathbf{x}(t) = \Phi(t)\mathbf{x}(0) + \int_0^t \Phi(t-\tau)\mathbf{B}\mathbf{u}(\tau)d\tau \quad (8.51)$$

In equation (8.51) the first term represents the response to a set of initial conditions, whilst the integral term represents the response to a forcing function.

### **Characteristic equation**

Using a state variable representation of a system, the characteristic equation is given by

$$|(s\mathbf{I} - \mathbf{A})| = 0 \quad (8.52)$$

## 8.2.1 Transient solution from a set of initial conditions

### Example 8.6

For the spring–mass–damper system given in Example 8.1, Figure 8.1, the state equations are shown in equation (8.13)

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -\frac{K}{m} & -\frac{C}{m} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{m} \end{bmatrix} u \quad (8.53)$$

Given:  $m = 1$  kg,  $C = 3$  Ns/m,  $K = 2$  N/m,  $u(t) = 0$ . Evaluate,

- the characteristic equation, its roots,  $\omega_n$  and  $\zeta$
- the transition matrices  $\phi(s)$  and  $\phi(t)$
- the transient response of the state variables from the set of initial conditions

$$\begin{aligned} y(0) &= 1.0, \\ \dot{y}(0) &= 0 \end{aligned}$$

### Solution

Since  $x_1 = y$  and  $x_2 = \dot{y}$ , then  $x_1(0) = 1.0$ ,  $x_2(0) = 0$ .

Inserting values of system parameters into equation (8.53) gives

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -2 & -3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u$$

$$(a) \quad (s\mathbf{I} - \mathbf{A}) = \begin{bmatrix} s & 0 \\ 0 & s \end{bmatrix} - \begin{bmatrix} 0 & 1 \\ -2 & -3 \end{bmatrix} = \begin{bmatrix} s & -1 \\ 2 & (s+3) \end{bmatrix} \quad (8.54)$$

From equation (8.52), the characteristic equation is

$$|(s\mathbf{I} - \mathbf{A})| = s(s+3) - (-2) = s^2 + 3s + 2 = 0 \quad (8.55)$$

Roots of characteristic equation

$$s = -1, -2 \quad (8.56)$$

Compare equation (8.55) with the denominator of the standard form in equation (3.43)

$$\begin{aligned} \omega_n^2 &= 2 \quad \text{i.e.} \quad \omega_n = 1.414 \text{ rad/s} \\ 2\zeta\omega_n &= 3 \quad \text{i.e.} \quad \zeta = 1.061 \end{aligned} \quad (8.57)$$

(b) The inverse of any matrix  $\mathbf{A}$  (see equation A2.17) is

$$\mathbf{A}^{-1} = \frac{\text{Adjoint } \mathbf{A}}{\det \mathbf{A}} \quad (8.58)$$

From equation (8.48)

$$\Phi(s) = (s\mathbf{I} - \mathbf{A})^{-1}$$

Using the standard matrix operations given in Appendix 2, equation (A2.12)

$$\text{Minors of } \Phi(s) = \begin{bmatrix} (s+3) & 2 \\ -1 & s \end{bmatrix}$$

$$\text{Co-factors of } \Phi(s) = \begin{bmatrix} (s+3) & -2 \\ 1 & s \end{bmatrix}$$

The Adjoint matrix is the transpose of the Co-factor matrix

$$\text{Adjoint of } \Phi(s) = \begin{bmatrix} (s+3) & 1 \\ -2 & s \end{bmatrix} \quad (8.59)$$

Hence, from equations (8.58) and (8.48)

$$\Phi(s) = \begin{bmatrix} \frac{(s+3)}{(s+1)(s+2)} & \frac{1}{(s+1)(s+2)} \\ \frac{-2}{(s+1)(s+2)} & \frac{s}{(s+1)(s+2)} \end{bmatrix} \quad (8.60)$$

Using partial fraction expansions

$$\Phi(s) = \begin{bmatrix} \left( \frac{2}{s+1} - \frac{1}{s+2} \right) & \left( \frac{1}{s+1} - \frac{1}{s+2} \right) \\ -2 \left( \frac{1}{s+1} - \frac{1}{s+2} \right) & \left( -\frac{1}{s+1} + \frac{2}{s+2} \right) \end{bmatrix} \quad (8.61)$$

Inverse transform equation (8.61)

$$\Phi(t) = \begin{bmatrix} (2e^{-t} - e^{-2t}) & (e^{-t} - e^{-2t}) \\ -2(e^{-t} - e^{-2t}) & (-e^{-t} + 2e^{-2t}) \end{bmatrix} \quad (8.62)$$

Note that the exponential indices are the roots of the characteristic equation (8.56).

(c) From equation (8.51), the transient response is given by

$$\mathbf{x}(t) = \Phi(t)\mathbf{x}(0) \quad (8.63)$$

Hence

$$\begin{bmatrix} x_1 \\ x_2 \end{bmatrix} = \begin{bmatrix} (2e^{-t} - e^{-2t}) & (e^{-t} - e^{-2t}) \\ -2(e^{-t} - e^{-2t}) & (-e^{-t} + 2e^{-2t}) \end{bmatrix} \begin{bmatrix} 1 \\ 0 \end{bmatrix} \quad (8.64)$$

$$x_1(t) = (2e^{-t} - e^{-2t}) \quad (8.65)$$

$$x_2(t) = -2(e^{-t} - e^{-2t})$$

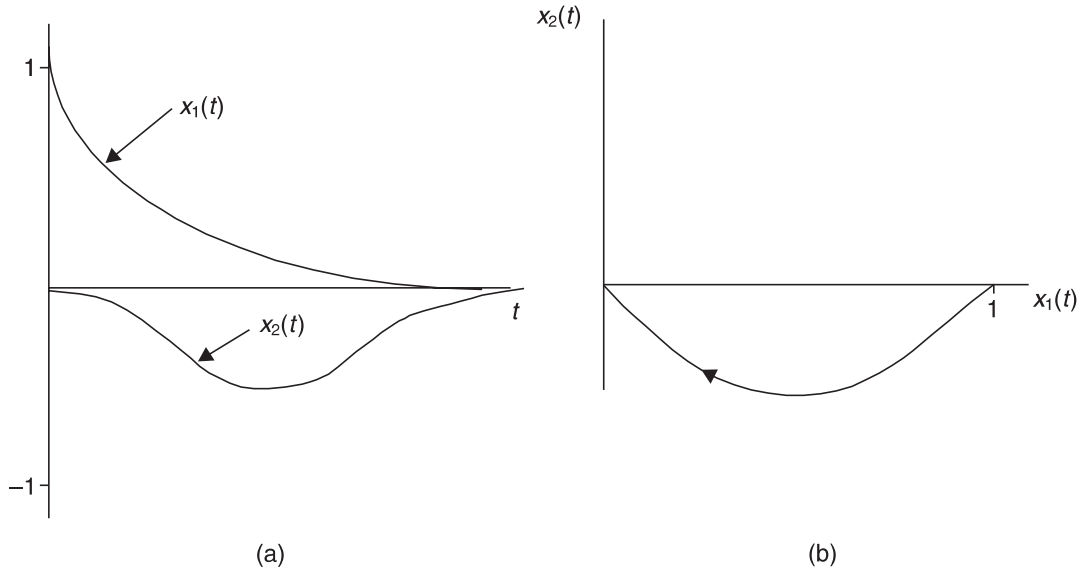
The time response of the state variables (i.e. position and velocity) together with the state trajectory is given in Figure 8.5.

### Example 8.7

For the spring–mass–damper system given in Example 8.6, evaluate the transient response of the state variables to a unit step input using

- The convolution integral
- Inverse Laplace transforms

Assume zero initial conditions.



**Fig. 8.5** State variable time response and state trajectory for Example 8.4.

*Solution*

(a) From equation (8.51)

$$\mathbf{x}(t) = \Phi(t) \begin{bmatrix} 0 \\ 0 \end{bmatrix} + \int_0^t \begin{bmatrix} \phi_{11}(t-\tau) & \phi_{12}(t-\tau) \\ \phi_{21}(t-\tau) & \phi_{22}(t-\tau) \end{bmatrix} \begin{bmatrix} 0 \\ \frac{1}{m} \end{bmatrix} \mathbf{u}(\tau) d\tau \quad (8.66)$$

Given that  $u(t) = 1$  and  $1/m = 1$ , equation (8.66) reduces to

$$\mathbf{x}(t) = \int_0^t \begin{bmatrix} \phi_{12}(t-\tau) \\ \phi_{22}(t-\tau) \end{bmatrix} d\tau$$

Inserting values from equation (8.62)

$$\mathbf{x}(t) = \int_0^t \begin{bmatrix} e^{-(t-\tau)} - e^{-2(t-\tau)} \\ e^{-(t-\tau)} + 2e^{-2(t-\tau)} \end{bmatrix} d\tau \quad (8.67)$$

Integrating

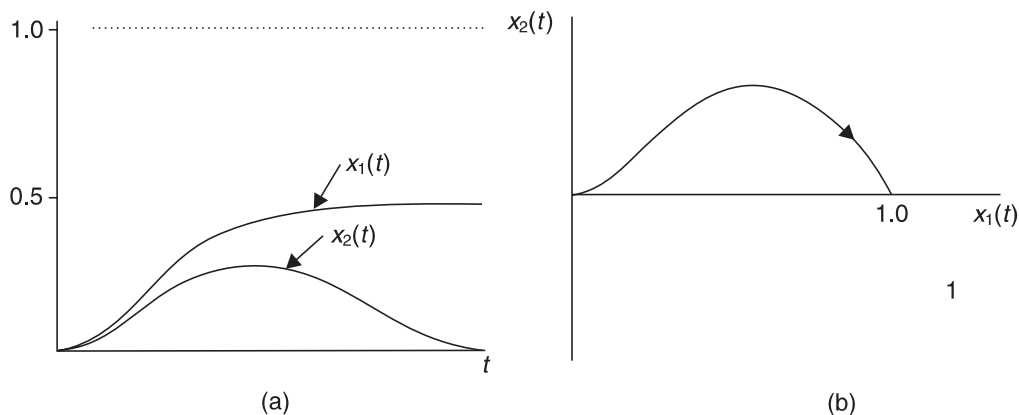
$$\mathbf{x}(t) = \begin{bmatrix} e^{-(t-\tau)} - \frac{1}{2}e^{-2(t-\tau)} \\ e^{-(t-\tau)} + e^{-2(t-\tau)} \end{bmatrix}_0^t \quad (8.68)$$

Inserting integration limits ( $\tau = t$  and  $\tau = 0$ )

$$\begin{bmatrix} x_1 \\ x_2 \end{bmatrix} = \begin{bmatrix} \frac{1}{2} - e^{-t} + \frac{1}{2}e^{-2t} \\ e^{-t} - e^{-2t} \end{bmatrix} \quad (8.69)$$

(b) An alternative method is to inverse transform from an  $s$ -domain expression. Equation (8.45) may be written

$$\mathbf{X}(s) = \Phi(s)\mathbf{x}(0) + \Phi(s)\mathbf{B}\mathbf{U}(s) \quad (8.70)$$



**Fig. 8.6** State variable step response and state trajectory for Example 8.5.

Hence from equation (8.61)

$$\mathbf{X}(s) = \Phi(s) \begin{bmatrix} 0 \\ 0 \end{bmatrix} + \begin{bmatrix} \left(\frac{2}{s+1} - \frac{1}{s+2}\right) & \left(\frac{1}{s+1} - \frac{1}{s+2}\right) \\ -2\left(\frac{1}{s+1} - \frac{1}{s+2}\right) & \left(\frac{-1}{s+1} + \frac{2}{s+2}\right) \end{bmatrix} \begin{bmatrix} 0 \\ 1 \end{bmatrix} \frac{1}{s} \quad (8.71)$$

Simplifying

$$\mathbf{X}(s) = \begin{bmatrix} \frac{1}{s(s+1)} - \frac{1}{2} \left\{ \frac{2}{s(s+2)} \right\} \\ \frac{-1}{s(s+1)} + \frac{2}{s(s+2)} \end{bmatrix} \quad (8.72)$$

Inverse transform

$$\mathbf{x}(t) = \begin{bmatrix} (1 - e^{-t}) - \frac{1}{2}(1 - e^{-2t}) \\ -(1 - e^{-t}) + (1 - e^{-2t}) \end{bmatrix} \quad (8.73)$$

which gives

$$\begin{bmatrix} x_1 \\ x_2 \end{bmatrix} = \begin{bmatrix} \frac{1}{2} - e^{-t} + \frac{1}{2}e^{-2t} \\ e^{-t} - e^{-2t} \end{bmatrix} \quad (8.74)$$

Equation (8.74) is the same as equation (8.69).

The step response of the state variables, together with the state trajectory, is shown in Figure 8.6.

### 8.3 Discrete-time solution of the state vector differential equation

The discrete-time solution of the state equation may be considered to be the vector equivalent of the scalar difference equation method developed from a *z*-transform approach in Chapter 7.